





## **SDI170**

## MEMS Quartz Tactical Inertial Measurement Unit

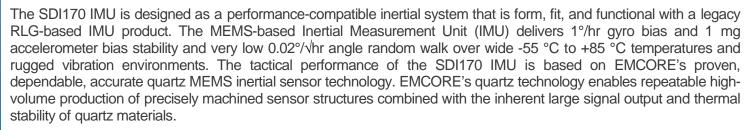
Suitable for Wide Variety of High-Precision Commercial, Industrial, **Marine and Defense Applications:** 

- Precision Aircraft Attitude Heading Reference Systems
- **GPS-Aided Navigation Systems**
- **Autonomous Vehicles**
- Remotely Operated Vehicles
- **Tactical Weapons & Torpedoes**
- Aerial and Marine Geomapping / Surveying
- Targeting & Pointing Systems
- Robotics

## **Key Performance Features:**

- 1°/hr Gyro & 1mg Accel Bias Over Wide Temperature Range
- 0.02°/vhr Angle Random Walk 5X Better Than RLG-based IMU
- < 1.0 Seconds Valid Data Start Up
- **Industry Standard RS-422 Serial Interface**
- < 33 in.3 Rugged Size
- Stable Performance, Superior Quality & Reliability
  - o 20 Year Lifetime Without Calibration
  - Greater Than 100.000 hr MTBF
- Designed as a Form, Fit & Functional Performance Alternative to Legacy RLG-based IMU





The SDI170 IMU is constructed with EMCORE's latest generation quartz gyros, quartz accelerometers, and high-speed signal processing to achieve outstanding precision performance. The SDI170 IMU is designed for demanding, missioncritical, rugged environments. It is ideal for continuous use applications with no wear-out components, highly linear accelerometer performance and longer life compared to competing alternatives. The solid-state quartz sensors and hermetically sealed IMU construction provide reliable MTBF and a 20-year operating and storage life. Continuous Builtin Test (BIT), configurable communications protocols, electromagnetic interference (EMI) protection and flexible input power compatibility make the SDI170 IMU easy to use in a wide range of higher-order integrated commercial and defense system applications.



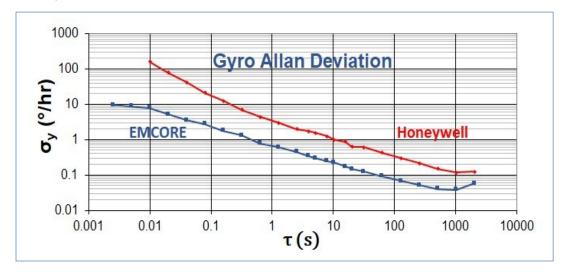




Specifications are based on 100 Hz Inertial Data ( $\Delta V/\Delta \theta$ )

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	Units	Measure	SDI170-AA00	SDI170-BA00	SDI170-CA00
System Performance					
Start Up Time for Valid Data	secs	max	1.0		
Bandwidth, Phase (-90° Phase Shift) *	Hz	min	90		
Gyro Channels					
Bias (over temperature)	deg/hr	1σ	1.0	2.0	3.0
Bias In-Run Stability	deg/hr	1σ	1.0	1.0	1.5
Scale Factor Error	ppm	1σ	200	200	200
Angle Random Walk	deg/√hr	max	0.02	0.02	0.02
Angular Rate – Dynamic Range	deg/sec	min	±1074	±1074	±1074
Accelerometer Channels					
Bias (over temperature)	milli-g	1σ	1.0	1.0	2.0
Bias In-Run Stability	μg	1σ	100	200	200
Scale Factor Error	ppm	1σ	200	200	200
Velocity Random Walk	μg/√Hz	1σ	100	100	120
Acceleration - Calibrated Range	g	min	±50 (±70 optional)	±50 (±70 optional)	±50 (±70 optional)
System Physical & Environmental					
Input Voltage	Vdc		+15 and +5		
Power	watts		<5.0		
I/O			RS422, SDLC		
Data Rate	Hz		100 Hz (Guidance) and 600 Hz (Control) – Other rates available		
Dimensions (height x diameter)	in		2.9 x 3.5		
Volume	cu in		<33		
Weight	lbs		<1.95		
Temperature	°C		-55 to +85		
Vibration (Operating)	g, rms		19		
Shock	g, ms		150, 11		
Operating Life	yrs		20		
Reliability @ 35°C (MTBF)	hrs		100,000 ground benign: 15,000 air inhabited cargo		
Dormancy	yrs		20		

Note: \* @ 600 Hz Flight Control Data Rate



## For more information, contact:

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